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## Appendix A

### Elements of the stiffness matrix

$$\bar{K}_{11} = A_{11}\alpha^2 + A_{66}\beta^2$$

$$\bar{K}_{12} = (A_{12} + A_{66})\alpha\beta$$

$$\bar{K}_{13} = -(B_{11}\alpha^3 + B_{12}\alpha\beta^2 + 2B_{66}\alpha\beta^2)$$

$$\bar{K}_{14} = \Omega_1 B_{11}\alpha^2 + C_{11}\alpha^2 + \Omega_1 B_{66}\beta^2 + C_{66}\beta^2$$

$$\bar{K}_{15} = (\Omega_2 B_{12} + D_{12} + \Omega_2 B_{66} + D_{66})\alpha\beta$$

$$\bar{K}_{31} = -(B_{11}\alpha^3 + B_{12}\alpha\beta^2 + 2B_{66}\alpha\beta^2)$$

$$\bar{K}_{32} = -(B_{22}\beta^3 + B_{12}\alpha^2\beta + 2B_{66}\alpha^2\beta)$$

$$\bar{K}_{33} = G_{11}\alpha^4 + 2G_{12}\alpha^2\beta^2 + 4G_{66}\alpha^2\beta^2 + G_{22}\beta^4$$

$$\bar{K}_{34} = -\left( \begin{array}{l} \Omega_1 G_{11}\alpha^3 + H_{11}\alpha^3 + 2\Omega_1 G_{66}\alpha\beta^2 \\ + 2H_{66}\alpha\beta^2 + \Omega_1 G_{12}\alpha\beta^2 + H_{12}\alpha\beta^2 \end{array} \right)$$

$$\bar{K}_{35} = -\left( \begin{array}{l} \Omega_2 G_{12}\alpha^2\beta + I_{12}\alpha^2\beta + 2\Omega_2 G_{66}\alpha^2\beta \\ + 2I_{66}\alpha^2\beta + \Omega_2 G_{22}\beta^3 + I_{22}\beta^3 \end{array} \right)$$

$$\bar{K}_{51} = (\Omega_2 B_{12} + D_{12} + \Omega_2 B_{66} + D_{66})\alpha\beta$$

$$\bar{K}_{52} = \Omega_2 B_{22}\beta^2 + D_{22}\beta^2 + \Omega_2 B_{66}\alpha^2 + D_{66}\alpha^2$$

$$\bar{K}_{53} = -\left( \begin{array}{l} \Omega_2 G_{12}\alpha^2\beta + I_{12}\alpha^2\beta + 2\Omega_2 G_{66}\alpha^2\beta \\ + 2I_{66}\alpha^2\beta + \Omega_2 G_{22}\beta^3 + I_{22}\beta^3 \end{array} \right)$$

$$\bar{K}_{54} = \left( \begin{array}{l} \Omega_1 \Omega_2 G_{12} + \Omega_1 I_{12} + \Omega_2 H_{12} + M_{12} \\ + \Omega_1 \Omega_2 G_{66} + \Omega_1 I_{66} + \Omega_2 H_{66} + M_{66} \end{array} \right) \alpha\beta$$

$$\bar{K}_{55} = \left( \begin{array}{l} \Omega_2^2 G_{22}\beta^2 + 2\Omega_2 I_{22}\beta^2 + P_{22}\beta^2 \\ + \Omega_2^2 G_{66}\alpha^2 + 2\Omega_2 I_{66}\alpha^2 + P_{66}\alpha^2 \\ + \Omega_2^2 AA_{11} + 2\Omega_2 EE_{11} + SS_{11} \end{array} \right)$$

$$\bar{K}_{21} = (A_{12} + A_{66})\alpha\beta$$

$$\bar{K}_{22} = A_{66}\alpha^2 + A_{22}\beta^2$$

$$\bar{K}_{23} = -(B_{22}\beta^3 + B_{12}\alpha^2\beta + 2B_{66}\alpha^2\beta)$$

$$\bar{K}_{24} = (\Omega_1 B_{66} + C_{66} + \Omega_1 B_{12} + C_{12})\alpha\beta$$

$$\bar{K}_{25} = \Omega_2 B_{22}\beta^2 + D_{22}\beta^2 + \Omega_2 B_{66}\alpha^2 + D_{66}\alpha^2$$

$$\bar{K}_{41} = \Omega_1 B_{11}\alpha^2 + C_{11}\alpha^2 + \Omega_1 B_{66}\beta^2 + C_{66}\beta^2$$

$$\bar{K}_{42} = (\Omega_1 B_{66} + C_{66} + \Omega_1 B_{12} + C_{12})\alpha\beta$$

$$\bar{K}_{43} = -\left( \begin{array}{l} \Omega_1 G_{11}\alpha^3 + H_{11}\alpha^3 + 2\Omega_1 G_{66}\alpha\beta^2 \\ + 2H_{66}\alpha\beta^2 + \Omega_1 G_{12}\alpha\beta^2 + H_{12}\alpha\beta^2 \end{array} \right)$$

$$\bar{K}_{44} = \left( \begin{array}{l} \Omega_1^2 G_{11}\alpha^2 + 2\Omega_1 H_{11}\alpha^2 + L_{11}\alpha^2 \\ + \Omega_1^2 G_{66}\beta^2 + 2\Omega_1 H_{66}\beta^2 + L_{66}\beta^2 \\ + \Omega_1^2 AA_{22} + 2\Omega_1 FF_{22} + UU_{22} \end{array} \right)$$

$$\bar{K}_{45} = \left( \begin{array}{l} \Omega_1 \Omega_2 G_{12} + \Omega_1 I_{12} + \Omega_2 H_{12} + M_{12} \\ + \Omega_1 \Omega_2 G_{66} + \Omega_1 I_{66} + \Omega_2 H_{66} + M_{66} \end{array} \right) \alpha\beta$$

$$[\bar{K}] = \begin{bmatrix} \bar{K}_{11} & \bar{K}_{12} & \bar{K}_{13} & \bar{K}_{14} & \bar{K}_{15} \\ \bar{K}_{21} & \bar{K}_{22} & \bar{K}_{23} & \bar{K}_{24} & \bar{K}_{25} \\ \bar{K}_{31} & \bar{K}_{32} & \bar{K}_{33} & \bar{K}_{34} & \bar{K}_{35} \\ \bar{K}_{41} & \bar{K}_{42} & \bar{K}_{43} & \bar{K}_{44} & \bar{K}_{45} \\ \bar{K}_{51} & \bar{K}_{52} & \bar{K}_{53} & \bar{K}_{54} & \bar{K}_{55} \end{bmatrix}$$



## Appendix B

### Elements of the mass matrix

$$\begin{aligned}\bar{M}_{11} &= \bar{I}_0; \bar{M}_{12} = 0; \bar{M}_{13} = -\bar{I}_1\alpha; \bar{M}_{14} = \bar{I}_3; \bar{M}_{15} = 0; \\ \bar{M}_{21} &= 0; \bar{M}_{22} = \bar{I}_0; \bar{M}_{23} = -\bar{I}_1\beta; \bar{M}_{24} = 0; \bar{M}_{25} = \bar{I}_6; \\ \bar{M}_{31} &= -\bar{I}_1\alpha; \bar{M}_{32} = -\bar{I}_1\beta; \bar{M}_{33} = \bar{I}_2(\alpha^2 + \beta^2) + \bar{I}_0; \\ \bar{M}_{34} &= -\bar{I}_4\alpha; \bar{M}_{35} = -\bar{I}_7\beta; \bar{M}_{41} = \bar{I}_3; \bar{M}_{42} = 0; \\ \bar{M}_{43} &= -\bar{I}_4\alpha; \bar{M}_{44} = \bar{I}_5; \bar{M}_{45} = 0; \\ \bar{M}_{51} &= 0; \bar{M}_{52} = \bar{I}_6; \bar{M}_{53} = -\bar{I}_7\beta; \bar{M}_{54} = 0; \bar{M}_{55} = \bar{I}_8\end{aligned}$$

$$[M] = \begin{bmatrix} \bar{M}_{11} & \bar{M}_{12} & \bar{M}_{13} & \bar{M}_{14} & \bar{M}_{15} \\ \bar{M}_{21} & \bar{M}_{22} & \bar{M}_{23} & \bar{M}_{24} & \bar{M}_{25} \\ \bar{M}_{31} & \bar{M}_{32} & \bar{M}_{33} & \bar{M}_{34} & \bar{M}_{35} \\ \bar{M}_{41} & \bar{M}_{42} & \bar{M}_{43} & \bar{M}_{44} & \bar{M}_{45} \\ \bar{M}_{51} & \bar{M}_{52} & \bar{M}_{53} & \bar{M}_{54} & \bar{M}_{55} \end{bmatrix}$$





## About the author



The author, Aniket Chanda, son of Shri Chandra Sekhar Chanda and Smt. Gopa Chanda, was born on 22<sup>nd</sup> June, 1992 at Navadwip, Nadia District, West Bengal. He graduated in Civil Engineering in the year 2015 from West Bengal University of Technology. Thereafter, he obtained his master's degree (M.Tech) in Civil Engineering in the year 2018 with a specialization in Structural Engineering from the Department of Civil Engineering, National Institute of Technology, Uttarakhand, India. In July 2018, he joined the Ph.D. program at Indian Institute of Technology (BHU), Varanasi in the Department of Civil Engineering, and this research work is carried out during this period.

## List of Publications from the Thesis

1. Chanda, A. and Sahoo, R., 2020. Accurate stress analysis of laminated composite and sandwich plates. *The Journal of Strain Analysis for Engineering Design*, 56(2), 96-111.
2. Chanda, A. and Sahoo, R., 2020. Analytical modeling of laminated composite plates integrated with piezoelectric layer using Trigonometric Zigzag theory. *Journal of Composite Materials*, 54(29), pp.4691-4708.
3. Chanda, A. and Sahoo, R., 2020. Flexural Behavior of Functionally Graded Plates with Piezoelectric Materials. *Arabian Journal for Science and Engineering*, 45(11), pp.9227-9248.
4. Chanda, A., Chandel, U., Sahoo, R. and Grover, N., 2020. Stress analysis of smart composite plate structures. *Proceedings of the Institution of Mechanical Engineers, Part C: Journal of Mechanical Engineering Science*, p.0954406220975449.
5. Chanda, A. and Sahoo, R., 2021. Static and dynamic responses of simply supported sandwich plates using non-polynomial zigzag theory. *Structures*, 29, pp. 1911-1933.

6. Chanda, A. and Sahoo, R., 2021. Trigonometric zigzag theory for free vibration and transient responses of cross-ply laminated composite plates. *Mechanics of Materials*, 155, p.103732.
7. Chanda, A. and Sahoo, R., 2021. Forced Vibration Responses of Smart Composite Plates using Trigonometric Zigzag Theory. *International Journal of Structural Stability and Dynamics*, p.2150067.
8. Sahoo, R. and Chanda, A., 2021. Transient analysis of smart composite laminate. *The Journal of Strain Analysis for Engineering Design*, 56(4), pp.225-248.
9. Chanda A and Sahoo R., 2021. Finite Element Analysis of smart composite plate structures coupled with piezoelectric materials: Investigation of Static and Vibration responses. *Mechanics of Advanced Materials and Structures* (Accepted)
10. Chanda A and Sahoo R. A study on the Stress and Vibration characteristics of laminated composite plates resting on Elastic Foundations using Analytical and Finite Element solutions. (Under Review)
11. Chanda A and Sahoo R. Analytical and Finite Element Models for the Static and Dynamic Electro-Elastic Responses of Smart Laminated Composite Plates with Piezoelectric Materials supported on elastic foundations. (Under Review)